Closed Form Solution of Nonlinear-Quadratic Optimal Control Problem by State-Control Parameterization using Chebyshev Polynomials

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ABSTRACT

In this paper the quasilinearization technique along with the Chebyshev polynomials of the first type are used to solve the nonlinear-quadratic optimal control problem with terminal state constraints. The quasilinearization is used to convert the nonlinear quadratic optimal control problem into sequence of linear quadratic optimal control problems. Then by approximating the state and control variables using Chebyshev polynomials, the optimal control problem can be approximated by a sequence of quadratic programming problems. The paper presents a closed form solution that relates the parameters of each of the quadratic programming problems to the original problem parameters. To illustrate the numerical behavior of the proposed method, the solution of the Van der Pol oscillator problem with and without terminal state constraints is presented.

General Terms:

Control Systems, Numerical Method

Keywords:

Nonlinear optimal control problem, Chebyshev polynomials, Quasilinearization, Iterative method

1. INTRODUCTION

Optimal control problems can be solved using the direct method by employing either the discretization technique or the parameterization technique [1-8]. There is a number of research papers which describe the use of the parameterization technique. Some accomplish parameterization by approximating the control variables by a function of unknown parameters [2], some by approximating the state variables [5-6,9] and some by approximating both the state and the control variables [3-4,7,10-11]

In [10], a method to solve a nonlinear optimal control problem by approximating the state and control variables by Chebyshev series has been proposed. The method converts the nonlinear optimal control problem into a nonlinear mathematical programming problem. The proposed method, however, is complicated in approximating the state equations and the performance index.

In [12], Jaddu presented a method to solve the constrained linear optimal control problem using state-control parameterization. Also in [8], a closed form solution of the linear quadratic optimal control problem was proposed. In this paper, the method of [12] and the work of [8] are extended to obtain a closed form solution of a nonlinear optimal control problem subject to initial and terminal state constraints. The proposed method is based on using the quasilinearization, therefore, the original nonlinear optimal control problem is approximated by a sequence of time-varying linear-quadratic optimal control problems. Then each of the state variables and control variables is approximated by a finite length Chebyshev series of unknown parameters. In this way the original nonlinear optimal control problem is converted into a sequence of quadratic programming problems. The use of the quasilinearization enables us to express the quadratic programming problems' parameters in terms of the parameters of the original problem, which simplifies the implementation of the method. The method approximates the state equations and the performance index in a different and easier way in comparison with the method of [10]. To illustrate the effectiveness of the proposed algorithm, the results of solving Van der Pol oscillator optimal control problem with and without terminal state constraints are presented.

2. PROBLEM STATEMENT

The system whose behaviour is described by the following system of differential equations is considered

$$\dot{\mathbf{x}} = \mathbf{f}(t, \mathbf{x}, \mathbf{u}) \qquad 0 \le t \le t_f \tag{1}$$

where t_f is a fixed terminal time; ${f x}$ is an n imes 1 state vector, ${f u}$ is an $m \times 1$ control vector; $\mathbf{f} : \mathbb{R}^n \times \mathbb{R}^m \to \mathbb{R}^n$ is continuously differentiable with respect to its arguments. The initial condition of the system is:

 $\mathbf{x}(0)$

$$) = \mathbf{x}_0 \tag{2}$$

and the terminal state constraints are given by:

$$\mathbf{x}(t_f) \in \mathcal{S}$$

where S is a given nonempty subset of \mathbb{R}^n . The following case is treated in this paper

$$\mathcal{S} = \{ \mathbf{x} \in R^n | \mathbf{E}\mathbf{x}(t_f) = \mathbf{x}_f \}$$
(3)

where \mathbf{E} is an $s \times n$ constant matrix, \mathbf{x}_f is a given $s \times 1$ vector of terminal conditions. However, the method is equally suitable for $S = R^n$, that is, for problems with no terminal constraints.

As a set of feasible controls, take the set \mathcal{U} consisting of all piece-wise continuous function $\mathbf{u} : [0, t_f] \to \mathbb{R}^m$ for which there exists a solution of (1) such that (2) and (3) are satisfied.

The problem is to find a feasible control that minimizes, over the set of all feasible controls, the quadratic performance index $J : U \to R^1$ given by:

$$J(\mathbf{u}) = \int_0^{t_f} (\mathbf{x}^T \mathbf{Q} \mathbf{x} + \mathbf{u}^T \mathbf{R} \mathbf{u}) dt$$
(4)

where \mathbf{Q} is an $n \times n$ positive semidefinite matrix, and \mathbf{R} is an $m \times m$ positive definite matrix.

3. PROBLEM REFORMULATION

To solve the problem of the previous section, the second method of the quasilinearization is applied [13]. The performance index is expanded up to the second order and the constraints are expanded up to the first order around a nominal state vector $\mathbf{x}^k(t)$ and a nominal control vector $\mathbf{u}^k(t)$.

The quasilinearization replaces the nonlinear optimal control problem (1)-(4) by a sequence of time-varying linear-quadratic optimal control problems given, for each $k = 0, 1, 2, \cdots$, as follows: Minimize:

 $J^{k+1}(\mathbf{u}^{k+1}) = \int_0^{t_f} ((\mathbf{x}^{k+1})^T \mathbf{Q} \mathbf{x}^{k+1} + (\mathbf{u}^{k+1})^T \mathbf{R} \mathbf{u}^{k+1}) dt$ (5)

subject to the linearized state equations, initial conditions and terminal state constraints given by:

$$\dot{\mathbf{x}}^{k+1} = \mathbf{A}^k(t)\mathbf{x}^{k+1} + \mathbf{B}^k(t)\mathbf{u}^{k+1} + \mathbf{h}^k(t) \qquad (6)$$

$$\mathbf{x}^{k+1}(0) = \mathbf{x}_0 \tag{7}$$

$$\mathbf{E}\mathbf{x}^{k+1}(t_f) = \mathbf{x}_f \tag{8}$$

where:

$$\begin{split} \mathbf{h}^{k}(t) &= \left. \mathbf{f}(t, \mathbf{x}^{(k)}, \mathbf{u}^{(k)}) - \mathbf{A}^{k}(t) \mathbf{x}^{(k)} - \mathbf{B}^{k}(t) \mathbf{u}^{(k)} \right. \\ \mathbf{A}^{k}(t) &= \left. \frac{\partial \mathbf{f}(t, \mathbf{x}, \mathbf{u})}{\partial \mathbf{x}} \right|_{\mathbf{x}^{k}, \mathbf{u}^{k}} \\ \mathbf{B}^{k}(t) &= \left. \frac{\partial \mathbf{f}(t, \mathbf{x}, \mathbf{u})}{\partial \mathbf{u}} \right|_{\mathbf{x}^{k}, \mathbf{u}^{k}} \end{split}$$

The next step in the proposed method is to convert each of the linear quadratic optimal control problems (5)-(8) into a quadratic programming problem by approximating each of the state variables and control variables by finite length Chebyshev polynomials of unknown parameters. The time interval $t \in [0, t_f]$ of the optimal control problem has to be changed to $\tau \in [-1, 1]$ because Chebyshev polynomials are orthogonal on this range. In the new setting $\tau = -1$ is the initial time and $\tau = 1$ is the terminal time. Expressing the problem (5)-(8) in terms of τ , gives

$$J^{k+1}(\mathbf{u}^{k+1}) = \frac{t_f}{2} \int_{-1}^{1} ((\mathbf{x}^{k+1})^T \mathbf{Q} \mathbf{x}^{k+1} + (\mathbf{u}^{k+1})^T \mathbf{R} \mathbf{u}^{k+1}) d\tau$$
(9)

subject to the constraints given by:

$$\frac{2}{t_f} \frac{d\mathbf{x}^{k+1}(\tau)}{d\tau} = \mathbf{A}^k(\tau)\mathbf{x}^{k+1} + \mathbf{B}^k(\tau)\mathbf{u}^{k+1} + \mathbf{h}^k(\tau)$$
(10)

$$\mathbf{x}^{k+1}(-1) = \mathbf{x}_0 \tag{11}$$

$$\mathbf{E}\mathbf{x}^{k+1}(1) = \mathbf{x}_0 \tag{12}$$

$$\mathbf{S}\mathbf{x}^{n+1}(1) = \mathbf{x}_f \tag{12}$$

Expanding each of the state variables $\mathbf{x}^{k+1}(\tau)$ and the control variables $\mathbf{u}^{k+1}(\tau)$ by a Chebyshev series of order N, results in

$$x_j^{k+1}(\tau) = \sum_{i=0}^{N} a_i^j (k+1) T_i(\tau) \qquad j = 1, 2, \cdots, n$$
 (13)

$$u_r^{k+1}(\tau) = \sum_{i=0}^N b_i^r(k+1)T_i(\tau) \qquad r = 1, 2, \cdots, m \qquad (14)$$

where $a_i^j(k+1)$ s and $b_i^r(k+1)$ s are (N+1)n and (N+1)munknown parameters respectively; $T_i(\tau)$ is the i-th Chebyshev polynomial.

These two systems of equations can be rewritten using the Kronecker product \otimes [14] as follows:

$$\mathbf{x}^{k+1}(\tau) = (\mathbf{T}^T(\tau) \otimes \mathbf{I}_n) \mathbf{a}_{k+1}$$
(15)

$$\mathbf{u}^{k+1}(\tau) = (\mathbf{T}^T(\tau) \otimes \mathbf{I}_m) \mathbf{b}_{k+1}$$
(16)

where:

$$\mathbf{a}_{k+1}^{T} = \begin{bmatrix} a_{0}^{1}(k+1) \ a_{0}^{2}(k+1) \cdots a_{0}^{n}(k+1) \ a_{1}^{1}(k+1) \cdots \\ a_{1}^{n}(k+1) \cdots a_{N}^{n}(k+1) \end{bmatrix}$$
$$\mathbf{b}_{k+1}^{T} = \begin{bmatrix} b_{0}^{1}(k+1) \ b_{0}^{2}(k+1) \cdots b_{0}^{m}(k+1) \ b_{1}^{1}(k+1) \cdots \\ b_{1}^{m}(k+1) \cdots b_{N}^{m}(k+1) \end{bmatrix}$$
$$\mathbf{\Gamma}^{T}(\tau) = \begin{bmatrix} T_{0}(\tau) \ T_{1}(\tau) \ \cdots \ T_{N}(\tau) \end{bmatrix}$$

and $\mathbf{I}_n, \mathbf{I}_m$ are $n \times n$ and $m \times m$ identity matrices respectively.

3.1 Performance Index Approximation

Substituting (15) and (16) into (9), gives:

$$J_{N}^{k+1}(\mathbf{a}_{k+1}, \mathbf{b}_{k+1}) = \frac{t_{f}}{2} \int_{-1}^{1} \mathbf{a}_{k+1}^{T}(\mathbf{T} \otimes \mathbf{I}_{n}) \mathbf{Q}(\mathbf{T}^{T} \otimes \mathbf{I}_{n}) \mathbf{a}_{k+1} + \mathbf{b}_{k+1}^{T}(\mathbf{T} \otimes \mathbf{I}_{m}) \mathbf{R}(\mathbf{T}^{T} \otimes \mathbf{I}_{m}) \mathbf{b}_{k+1} d\tau$$
(17)

where the values of J_N^{k+1} are the approximate values of J^{k+1} obtained through approximation of the state and control variables by N-th order Chebyshev series. This equation can be simplified to:

$$J_N^{k+1}(\mathbf{a}_{k+1}, \mathbf{b}_{k+1}) = \frac{t_f}{2} \int_{-1}^{1} (\mathbf{a}_{k+1}^T (\mathbf{T}\mathbf{T}^T \otimes \mathbf{Q}) \mathbf{a}_{k+1} + \mathbf{b}_{k+1}^T (\mathbf{T}\mathbf{T}^T \otimes \mathbf{R}) \mathbf{b}_{k+1}) d\tau$$
(18)

which can be written in compact form as:

$$J_N^{k+1}(\mathbf{z}_{k+1}) = \frac{1}{2} \mathbf{z}_{k+1}^T \mathbf{H} \mathbf{z}_{k+1}$$
(19)

where:

$$\mathbf{z}_{k+1}^{T} = \begin{bmatrix} \mathbf{a}_{k+1}^{T} & \mathbf{b}_{k+1}^{T} \end{bmatrix}$$
$$\mathbf{H} = t_{f} \int_{-1}^{1} \begin{bmatrix} \mathbf{T}\mathbf{T}^{T} \otimes \mathbf{Q} & \mathbf{O} \\ \mathbf{O}^{T} & \mathbf{T}\mathbf{T}^{T} \otimes \mathbf{R} \end{bmatrix} d\tau$$

where **O** is $n(N + 1) \times m(N + 1)$ zero matrix. **H** is $(m+n)(N+1) \times (m+n)(N+1)$ matrix. The integration in this equation is performed element-wise.

3.2 Constraints Approximation

To approximate the state equation, the initial condition and the terminal state constraints, equation (13) is written as follows:

$$\mathbf{x}^{k+1} = \sum_{i=0}^{N} T_i \alpha_i (k+1)$$
(20)

or:

$$\mathbf{x}^{k+1} = \mathbf{T}^T [\alpha_0^T(k+1) \,\alpha_1^T(k+1) \cdots \alpha_N^T(k+1)]^T$$
$$= \mathbf{T}^T \alpha_{k+1}$$
(21)

where:

$$\alpha_i^T(k+1) = \left[a_i^1(k+1) \ a_i^2(k+1) \ \cdots \ a_i^n(k+1) \right]$$

Similarly, the control variables (14) can be written as:

$$\mathbf{u}^{k+1} = \mathbf{T}^{T} [\beta_{0}^{T}(k+1) \ \beta_{1}^{T}(k+1) \ \cdots \ \beta_{N}^{T}(k+1)]^{T} = \mathbf{T}^{T} \beta_{k+1}$$
(22)

where:

$$\beta_i^T(k+1) = \left[b_i^1(k+1) \ b_i^2(k+1) \ \cdots \ b_i^m(k+1) \right]$$

Notice that $\alpha_{k+1} = \mathbf{a}_{k+1}$ and $\beta_{k+1} = \mathbf{b}_{k+1}$. However, the multiplication in equations (21) and (22) has to take place block-wise, while the multiplication in equations (15) and (16) is performed element-wise.

Also, $\mathbf{A}^k(\tau)$, $\mathbf{B}^k(\tau)$ and $\mathbf{h}^k(\tau)$ have to be expanded by a finite Chebyshev series.

The approximation of $\mathbf{A}^k(\tau)$ can be given by:

$$\mathbf{A}^{k}(\tau) = \sum_{i=0}^{N} \mathbf{A}_{i} T_{i}$$
$$= [\mathbf{A}_{0} \ \mathbf{A}_{1} \ \cdots \ \mathbf{A}_{N}] \mathbf{T}$$
(23)

where \mathbf{A}_{i} , $i = 0, 1, \dots N$ is an $n \times n$ constant matrix of the coefficient of the Chebyshev polynomial T_i . \mathbf{A}_i 's can be obtained as follows [15]:

$$\mathbf{A}_{0} = \frac{1}{M} \sum_{n=1}^{M} \mathbf{A}(\cos(\theta_{n}))$$
$$\mathbf{A}_{i} = \frac{2}{M} \sum_{n=1}^{M} \mathbf{A}(\cos(\theta_{n}))\cos(i\theta_{n}) \quad i = 1, 2, \cdots, N$$

where $\theta_n = \frac{2n-1}{2M}\pi$, $n = 1, 2, \dots, M$ and M > N. Similarly, $\mathbf{B}^k(\tau)$ can be expanded by Chebyshev series and can be

Similarly, $\mathbf{B}^{\kappa}(\tau)$ can be expanded by Chebyshev series and can be expressed as:

$$\mathbf{B}^{k}(\tau) = [\mathbf{B}_{0} \ \mathbf{B}_{1} \ \cdots \ \mathbf{B}_{N}]\mathbf{T}$$
(24)

where \mathbf{B}_i is an $n \times m$ matrix of constant elements.

Following the same procedure the vector $\mathbf{h}^k(\tau)$ can be expanded into a Chebyshev series as:

$$\mathbf{h}^{k}(\tau) = \mathbf{T}^{T} [\mathbf{h}_{0}^{T} \mathbf{h}_{1}^{T} \cdots \mathbf{h}_{N}^{T}]^{T}$$
$$= \mathbf{T}^{T} \mathcal{H}$$
(25)

where:

$$\mathbf{h}_i^T = \left[\begin{array}{ccc} h_i^1 & h_i^2 & \cdots & h_i^n \end{array} \right]$$

Notice that the multiplications in (23), (24) and (25) is performed block-wise.

The last part of the state equation (10) to be approximated is $\dot{\mathbf{x}}^{k+1}$, which can be expressed in terms of the parameters of \mathbf{x}^{k+1} . This can be achieved, by using the differentiation operational matrix **D** [16] given in the appendix, as follows:

$$\dot{\mathbf{x}}^{k+1}(\tau) = \mathbf{T}^T \mathbf{D}^T \alpha_{k+1} \tag{26}$$

Substituting (21), (22), (23), (24), (25) and (26) into (10), gives

$$\frac{2}{t_f} \mathbf{T}^T \mathbf{D}^T \alpha_{k+1} = [\mathbf{A}_0 \cdots \mathbf{A}_N] \mathbf{T} \mathbf{T}^T \alpha_{k+1} + [\mathbf{B}_0 \cdots \mathbf{B}_N] \mathbf{T} \mathbf{T}^T \beta_{k+1} + \mathbf{T}^T \mathcal{H} \quad (27)$$

The right hand side can be simplified using the property of Chebyshev polynomials derived in [17]. Although this property is derived for the Chebyshev polynomials of the second type, it can be proved that the same property holds for the Chebyshev polynomials of the first type that are used in this work. Applying this property yields,

$$\left[\mathbf{A}_{0} \, \mathbf{A}_{1} \, \cdots \, \mathbf{A}_{N} \right] \mathbf{T} \mathbf{T}^{T} = \mathbf{T}^{T} \mathcal{A}$$
(28)

$$\left[\mathbf{B}_0 \, \mathbf{B}_1 \, \cdots \, \mathbf{B}_N \, \right] \mathbf{T} \mathbf{T}^T \; = \; \mathbf{T}^T \mathcal{B} \tag{29}$$

where \mathcal{A}, \mathcal{B} are constant matrices given in the appendix. Substituting (28) and (29) into (27) gives:

$$\frac{2}{t_f} \mathbf{T}^T \mathbf{D}^T \alpha_{k+1} = \mathbf{T}^T \mathcal{A} \alpha_{k+1} + \mathbf{T}^T \mathcal{B} \beta_{k+1} + \mathbf{T}^T \mathcal{H} \qquad (30)$$

In this equation the multiplications have to be performed block-wise. To be able to perform element-wise multiplication, this equation can be rewritten as:

$$\frac{2}{t_f} (\mathbf{T}^T \mathbf{D}^T \otimes \mathbf{I}_n) \alpha_{k+1} = (\mathbf{T}^T \otimes \mathbf{I}_n) \mathcal{A} \alpha_{k+1} + (\mathbf{T}^T \otimes \mathbf{I}_n) \mathcal{B} \beta_{k+1} + (\mathbf{T}^T \otimes \mathbf{I}_n) \mathcal{H} \quad (31)$$

Using the Kronecker product properties [14], this equation can written as:

$$\frac{2}{t_f} (\mathbf{T}^T \otimes \mathbf{I}_n) (\mathbf{D}^T \otimes \mathbf{I}_n) \alpha_{k+1} = (\mathbf{T}^T \otimes \mathbf{I}_n) \mathcal{A} \alpha_{k+1} + (\mathbf{T}^T \otimes \mathbf{I}_n) \mathcal{B} \beta_{k+1} + (\mathbf{T}^T \otimes \mathbf{I}_n) \mathcal{H} \quad (32)$$

Equating the coefficients of $\mathbf{T}^T \otimes \mathbf{I}_n$ on both sides, yields:

$$\frac{2}{t_f} (\mathbf{D}^T \otimes \mathbf{I}_n) \alpha_{k+1} = \mathcal{A} \alpha_{k+1} + \mathcal{B} \beta_{k+1} + \mathcal{H}$$
(33)

In this equation α_{k+1} and β_{k+1} can be replaced by \mathbf{a}_{k+1} and \mathbf{b}_{k+1} respectively to become:

$$\left(\mathcal{A} - \frac{2}{t_f} (\mathbf{D}^T \otimes \mathbf{I}_n)\right) \mathbf{a}_{k+1} + \mathcal{B} \mathbf{b}_{k+1} + \mathcal{H} = 0 \qquad (34)$$

This equation will replace the system state equation (10).

In addition to the state equations, the initial conditions and the terminal state constraints have to be approximated. Substituting (15) into both (11) and (12), yields:

$$(\mathbf{T}^T(-1) \otimes \mathbf{I}_n)\mathbf{a}_{k+1} = \mathbf{x}_0 \tag{35}$$

$$\mathbf{E}(\mathbf{T}^T(1) \otimes \mathbf{I}_n)\mathbf{a}_{k+1} = \mathbf{x}_f \tag{36}$$

Combining equations (35),(36) with (34), yields:

$$\begin{bmatrix} \mathcal{A} - \frac{2}{t_f} (\mathbf{D}^T \otimes \mathbf{I}_n) \\ \mathbf{T}^T (-1) \otimes \mathbf{I}_n \\ \mathbf{E} (\mathbf{T}^T (1) \otimes \mathbf{I}_n) \end{bmatrix} \mathbf{a}_{k+1} + \begin{bmatrix} \mathcal{B} \\ \mathbf{O}_{(n \times m(N+1))} \\ \mathbf{O}_{(s \times m(N+1))} \end{bmatrix} \mathbf{b}_{k+1} + \begin{bmatrix} \mathcal{H} \\ -\mathbf{x}_0 \\ -\mathbf{x}_f \end{bmatrix} = 0$$
(37)

or

$$\begin{bmatrix} \mathcal{A} - \frac{2}{t_f} (\mathbf{D}^T \otimes \mathbf{I}_n) & \mathcal{B} \\ \mathbf{T}^T (-1) \otimes \mathbf{I}_n & \mathbf{O}_{(n \times m(N+1))} \\ \mathbf{E} (\mathbf{T}^T (1) \otimes \mathbf{I}_n) & \mathbf{O}_{(s \times m(N+1))} \end{bmatrix} \mathbf{z}_{k+1} = \tilde{\mathbf{h}}$$
(38)

where

$$\mathbf{z}_{k+1} = \begin{bmatrix} \mathbf{a}_{k+1} \\ \mathbf{b}_{k+1} \end{bmatrix}$$
 $ilde{\mathbf{h}} = \begin{bmatrix} -\mathcal{H} \\ \mathbf{x}_0 \\ \mathbf{x}_f \end{bmatrix}$

Equation (38) represent the equality constraints that replace the state equations, initial condition and the terminal state constraints. Each of the time-varying linear-quadratic optimal control problems (5)-(8) is converted into a quadratic programming problem. The new problem is to find a vector \mathbf{z}_{k+1}^* that minimizes:

$$J_N^{k+1}(\mathbf{z}_{k+1}) = \frac{1}{2} \mathbf{z}_{k+1}^T \mathbf{H} \mathbf{z}_{k+1}$$
(39)

subject to:

$$\mathbf{F}\mathbf{z}_{k+1} = \tilde{\mathbf{h}} \tag{40}$$

where:

$$\mathbf{F} = \begin{bmatrix} \mathcal{A} - \frac{2}{t_f} (\mathbf{D}^T \otimes \mathbf{I}_n) & \mathcal{B} \\ \mathbf{T}(-1) \otimes \mathbf{I}_n & \mathbf{O}_{(n \times m(N+1))} \\ \mathbf{ET}(1) \otimes \mathbf{I}_n & \mathbf{O}_{(s \times m(N+1))} \end{bmatrix}$$

The *n* state equations are replaced by n(N+1) equality constraints while the initial condition and the terminal state constraints represent an additional n+s equality constraints. Hence each of the linear quadratic optimal control problems (9)- (12) is approximated by quadratic programming problem of (n+m)(N+1) unknown parameters and n(N+2) + s equality constraints. Therefore, to make sure that the number of the unknown parameters is greater than the number of equality constraints, the following inequality should hold when choosing the order N of the approximation:

$$N > \frac{n+s-m}{m} \tag{41}$$

The solution of the quadratic programming problem can be obtained by matrix-vector multiplication and the optimal value of the unknown parameters vector z is given by:

$$\mathbf{z}_{k+1}^* = \mathbf{H}^{-1} \mathbf{F}^T (\mathbf{F} \mathbf{H}^{-1} \mathbf{F}^T)^{-1} \tilde{\mathbf{h}}$$
(42)

After obtaining the optimal value of the unknown parameters vector, these values will be substituted back into equations (13) and (14) to obtain the new nominal state vector and the new nominal control vector that will be used to obtain the next linear quadratic optimal control problem. This process is continued until

 $|J_N^{k+1}-J_N^k|$ is small enough. In this work the computations are terminated when $|J_N^{k+1}-J_N^k|\leq 1\times 10^{-4}.$ In order to decide whether the computed solution is close enough

In order to decide whether the computed solution is close enough to the optimal solution, the following criteria is used: Substitution of the calculated control $\mathbf{u}^k(t)$ of the last iteration into the state equation (1) gives:

$$\dot{\mathbf{x}} = \mathbf{f}(t, \mathbf{x}(t), \mathbf{u}^k(t)) \tag{43}$$

Numerical integration of (43) is possible for a given initial or final conditions. If $\hat{\mathbf{x}}(t)$ is the solution of the numerical integration, then the following criteria can be used to estimate the error

$$\epsilon_{dyn} = \max_{0 \le t \le t_f} |\hat{\mathbf{x}}(t) - \mathbf{x}^k(t)| \tag{44}$$

4. NUMERICAL EXAMPLE

To illustrate the numerical behaviour of the proposed method, the Van der Pol oscillator problem is considered. The problem is to minimize the performance index:

$$J = \int_0^5 (x_1^2 + x_2^2 + u^2) dt$$
 (45)

subject to the state equations and initial conditions given by:

$$\dot{x}_1 = x_2 \tag{46}$$

$$\dot{x}_2 = -x_1 + x_2 - x_1^2 x_2 + u \tag{47}$$

 $x_1(0) = 1$ and $x_2(0) = 0$,

Applying the quasilinearization and expressing the problem in terms of τ , the following sequence of problems is obtained: For $k = 0, 1, 2 \cdots$, minimize:

$$J^{k+1} = \frac{5}{2} \int_{-1}^{1} \left((x_1^{k+1})^2 + (x_2^{k+1})^2 + (u^{k+1})^2 \right) d\tau \qquad (48)$$

subject to the linearized state equations and initial conditions:

$$\begin{array}{l} \frac{2}{5} \left[\begin{array}{c} \frac{dx_1^{k+1}}{d\tau} \\ \frac{dx_2^{k+1}}{d\tau} \end{array} \right] \; = \; \left[\begin{array}{cc} 0 & 1 \\ -1 - 2x_1^k x_2^k & 1 - x_1^{k} \end{array} \right] \left[\begin{array}{c} x_1^{k+1} \\ x_2^{k+1} \end{array} \right] + \left[\begin{array}{c} 0 \\ 1 \end{array} \right] u^{k+1} \\ & + \left[\begin{array}{c} 0 \\ 2x_1^{k^2} x_2^k \end{array} \right] \\ x_1^{k+1}(-1) \; = \; 1 \\ x_2^{k+1}(-1) \; = \; 0 \end{array}$$

This problem is solved using the proposed algorithm, starting from $x_1^0 = x_2^0 = 0$ and expanding each of the state variables and the control variable by Chebyshev series of unknown parameters. The cases considered are N = 3, N = 5, N = 7, N = 9 and N = 11. The resulting approximate optimal values, their differences in subsequent iterations and the error estimate ϵ_{dyn} are shown in Table 1. From this table, it is clear that a decreasing sequence of optimal values, with very satisfactory behaviour, is obtained as N increases.

The optimal states and the optimal control for N=11 are shown in Fig. 1 and Fig. 2 respectively.



Fig. 1. Optimal state x(t) for unconstrained case



Fig. 2. Optimal control u(t) for unconstrained case

Also, the previous problem is solved, but with the following terminal state constraints:

$$x_1(5) = -0.97$$

 $x_2(5) = -0.96$

for N = 5, N = 7, N = 9 and N = 11. The case, N = 3 is excluded because it does not satisfy the condition of (41). The approximate optimal values along with their differences in subsequent iterations and the error estimate ϵ_{dyn} are summarized

in Table 2. The initial and the terminal state constraints are exactly

satisfied. The optimal states and control for N = 11 are shown in Fig. 3 and Fig. 4 respectively.



Fig. 3. Optimal state x(t) for constrained case



Fig. 4. Optimal control u(t) for constrained case

The problem with terminal state constraints was solved by Frick and Stech [11] using Epsilon-Ritz method on parallel processor array. They found an approximate optimal value to be 4.2490 in five iterations.

By looking closely at Table 1 and Table 2, it is clear that for each N, an acceptable approximation of the optimal value is obtained for k = 2. By increasing k the optimal value does not change considerably as Table 3 shows. This table shows that the difference between the third and the second iteration is very small, and accurate results can be obtained from the second iteration. It is believed that this fast convergence is due to the use of Chebyshev polynomials in combination with the quasilinearization method.

N	k	J_{N}^{k+1}	$ J_N^{k+1} - J_N^k $	Edam
N=3	0	4.02168387	N	agn
	1	3.53249140	0.489192	
	2	3.52811178	0.004379	
	3	3.52827537	0.000163	
	4	3.52826923	0.000006	0.38
N=5	0	3.38355087	-	
	1	2.86896807	0.514580	
	2	2.86865514	0.000131	
	3	2.86867141	0.000016	0.032
N=7	0	3.37167025	_	
	1	2.86798210	0.503688	
	2	2.86789913	0.000082	1.8×10^{-3}
N = 9	0	3.37164338	-	
	1	2.86693354	0.504709	
	2	2.86703264	0.000099	6×10^{-4}
N=11	0	3.37164327	-	
	1	2.86685508	0.504788	
	2	2.86695986	0.000104	
	3	2.86695565	0.000004	7.3×10^{-5}

Table 1. Approximate optimal value for the first case

Table 2. Approximate optimal value for the second case

N	k	J_N^{k+1}	$\mid J_{N}^{k+1} - J_{N}^{k} \mid$	ϵ_{dyn}
N=5	0	4.49271577	_	
	1	4.24763913	0.245076	
	2	4.24842816	0.000789	
	3	4.24833707	0.000091	0.033
N=7	0	4.49107754	_	
	1	4.22138666	0.269690	
	2	4.22217961	0.000792	
	3	4.22219794	0.000018	$6.1 imes 10^{-3}$
N=9	0	4.49100236	_	
	1	4.22037466	0.270627	
	2	4.22015874	0.000215	
	3	4.22016087	0.000002	$0.83 imes10^{-3}$
N=11	0	4.49100235	_	
	1	4.22023610	0.270766	
	2	4.22004433	0.000191	
	3	4.22004500	0.0000006	$1.2 imes 10^{-4}$

Table 3. Difference between J_N^2 and J_N^3

unco	onstrained case	constrained case			
N	$ J_N^2 - J_N^3 $	N	$ J_N^2 - J_N^3 $		
3	0.000163				
5	0.000016	5	0.000091		
7	0.000003	7	0.000018		
9	0.000007	9	0.000002		
11	0.000004	11	0.0000006		

5. CONCLUSION

A computational method is proposed to solve the nonlinear quadratic optimal control problem with initial and terminal state constraints. The method reduces the problem into solving sequence of quadratic programming problems which can be solved easily by matrix vector multiplications. The sequence of the approximate optimal values generated by the method seems to converge very fast to the optimal value. It is believe that this is due to the use of quasilinearization and the use of Chebyshev polynomials.

6. APPENDIX

Chebyshev polynomials' differentiation operational matrix \mathbf{D} [16] is given by:

	$\begin{bmatrix} 0\\1 \end{bmatrix}$	$\begin{array}{c} 0 \\ 0 \end{array}$	· · ·	0 = 0							
	0	4	0	0	0	0	0	0		0	
	3	0	6	0	0	0	0	0		0	
	0	8	0	8	0	0	0	0		0	
$\mathbf{D} =$	5	0	10	0	10	0	0	0		0	(49)
	0	12	0	12	0	12	0	0		0	
	7	0	14	0	14	0	14	0	• • •	0	
	0	16	0	16	0	16	0	16	• • •	0	
	:	÷	÷	÷	÷	÷	÷	÷	÷	÷	

The matrix \mathcal{A} is given by:

$$\mathcal{A} = \begin{bmatrix} \mathbf{A}_{0} & \frac{\mathbf{A}_{1}}{2} & \frac{\mathbf{A}_{2}}{2} & \dots & \frac{\mathbf{A}_{N}}{2} \\ \mathbf{A}_{1} & \mathbf{A}_{0} + \frac{\mathbf{A}_{2}}{2} & \frac{(\mathbf{A}_{1} + \mathbf{A}_{3})}{2} & \dots & \frac{\mathbf{A}_{N-1}}{2} \\ \mathbf{A}_{2} & \frac{(\mathbf{A}_{1} + \mathbf{A}_{3})}{2} & \mathbf{A}_{0} + \frac{\mathbf{A}_{4}}{2} & \dots & \frac{\mathbf{A}_{N-2}}{2} \\ \mathbf{A}_{3} & \frac{(\mathbf{A}_{2} + \mathbf{A}_{4})}{2} & \frac{(\mathbf{A}_{1} + \mathbf{A}_{5})}{2} & \dots & \frac{\mathbf{A}_{N-3}}{2} \\ \vdots & \vdots & \vdots & \dots & \vdots \\ \mathbf{A}_{N} & \frac{\mathbf{A}_{N-1}}{2} & \frac{\mathbf{A}_{N-2}}{2} & \dots & \mathbf{A}_{0} \end{bmatrix}$$
(50)

The matrix \mathcal{B} is defined similarly.

7. REFERENCES

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