# System Simulation of RF Front-End Transceiver for Frequency Modulated Continuous Wave Radar

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# ABSTRACT

Frequency modulated continuous wave (FMCW) radar is a solution for range and velocity measurement of a target. This paper introduces a simulation methodology for a RF front-end transceiver of a FMCW radar system including the target model. The proposed system model can incorporate realistic design conditions such as voltage controlled oscillator phase noise. The methodology has been applied to a 2.45 GHz FMCW radar with a single antenna for transmitting and receiving waves using Advanced Design System (ADS) simulation package from Agilent. Various values of target range and velocity are employed in the simulated model to verify the operation of the FMCW radar system. The proposed simulation model has been proven to be a useful tool in the design of FMCW radar systems. The system model is used to conclude some general guidelines to help the system designer.

# **Keywords**

FMCW radar, Velocity, Doppler, Circulator, LNA.

# **1. INTRODUCTION**

Radar systems are being used in many applications nowadays. There are different types of radar systems; one of the wellknown radar systems is FMCW. FMCW radar systems have applications in various fields ranging from civil to military applications. FMCW radars can be used for imaging purposes ([1]-[4]); automotive FMCW radar can be used to record velocity violations on roads [5,6]. They can be also used as driver assistance systems to improve driving conditions and avoid collisions [7]. They can be also employed for ship navigation and identification as proposed by Duarte et. al. [8]. FMCW radars can be used for target detection under the ground clutter environment [9]. Some authors presented FMCW radars for geosciences to measure wind speed and direction [10] or to perform surface analysis to planets [11]. On the other hand, it is frequently used for military purposes to monitor and to guide airplanes and missiles [12].

In the frequency-modulation method, the transmitter radiates radio-frequency waves. The frequency of these RF waves is continually increasing and decreasing from a fixed reference frequency. At any instant, the frequency of the returned (reflected) signal differs from the frequency of the radiated one. The amount of the difference in the frequency is determined by the time it took the signal to travel the distance from the transmitter to the target [13]. Ahmed M. Nahhas

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In the system design of the RF front-end transceiver, a designer uses Friis equation and IP3 cascaded system equation for noise and nonlinearity estimate. The designer employs these equations to distribute the system specifications to the individual receiver blocks. Aleksandar et. al. [14] introduced a procedure for optimal allocation of the performance parameters to the individual RF front-end circuit blocks based on these two equations. However, there should be a simulating methodology to gain exactness of the system design as these two equations have their limitations. For the first time, to the best of our knowledge, a system level simulation methodology is proposed and thoroughly explained. The model of the target is included; the VCO phase noise has been also modeled in the simulation. Other effects, although not included, can be easily incorporated in the simulation. For example, one can model the attenuation and noise carried by the channel between the antenna and the target model.

The operation principle of the FMCW radar is discussed in section 2. Section 3 presents the generation of the FM chirp waveform and the frequency deviation relationship to the target range. Section 4 describes the development of the target model while section 5 explains the whole system model supported with simulation results. Some general design guidelines are withdrawn for the system designer. The article is finally concluded in section 6.

# 2. PRINCIPLE of FMCW RADAR

The carrier signal of the radar is frequency-modulated in linear ramps. The antenna of the radar transmits and receives simultaneously with a frequency deviation ( $\Delta F$ ) for both the transmitted and received signals, as shown in figure 1. The frequency difference ( $F_r$ ) is proportional to the time difference between the transmitted and received signals, which in turn is proportional to the distance between the transmitter and the reflecting-object (target).

In a regular FMCW radar, the frequency sweep over  $\Delta F$  Hz in T second at the carrier frequency is transmitted. The receiver mixes the transmitted signal with the delayed received signal; the output of the mixer constitutes the frequency difference and sum components. The mixer is followed by a low pass filter in order to pass the frequency difference component only. This is referred to a beat note whose frequency  $F_r$  is proportional to the range R of a stationary target or a target whose velocity produces a Doppler shift larger than  $\pm 1/2T$  Hz [15].

Assume that the transmitter frequency increases linearly with time and that there is a reflecting object at a distance  $r = r_o$ , A frequency modulated carrier signal can be written as:



Fig 1: Range determination of stationary target

$$\mathbf{u}_{s} = \mathbf{A}_{s} \cdot \cos \boldsymbol{\varphi}_{s}(\mathbf{t}) \tag{1}$$

$$f_s(t) = f_o + \frac{\Delta F}{T} \cdot t$$
 for  $0 \prec t \prec T$  (2)

$$\Phi_{s}(t) = 2\pi \int_{0}^{t} f_{s}(t)dt + \varphi_{so} = 2\pi \left(f_{o}t + \frac{1}{2} \cdot \frac{\Delta F}{T}t^{2}\right) + \varphi_{so}$$
(3)

where  $A_s$  is the signal amplitude,  $f_o$  the carrier frequency,  $\phi_{so}$  the initial phase, and  $f_s(t)$  a time varying frequency. An echo signal will return after the transit time  $\tau$ . The phase of the received signal can be expressed as:

$$\varphi_{E}(t) = \varphi_{s}(t-\tau) = 2\pi \left(f_{o}(t-\tau) + \frac{1}{2} \cdot \frac{\Delta F}{T}(t-\tau)^{2}\right) - \varphi_{so} \quad (4)$$

If the echo signal is heterodyned with a portion of the transmitter signal in a nonlinear element such as a diode, a beat note will be produced. The beat note component can be written as:

$$u_{d}(t) = U_{D}\cos(\varphi_{s}(t) - \varphi_{E}(t))$$
(5)

$$u_{d}(t) = U_{D} \cos \left[ 2\pi \left[ \frac{\Delta F}{T} t\tau + f_{o} \tau - \frac{1}{2} \Delta F \tau \frac{\tau}{T} \right] \right]$$
(6)

where  $U_D$  is the amplitude of the beat note signal. If there is no Doppler frequency shift (non moving target), the beat note (difference frequency) is a measure of the target's range.

$$\tau = \frac{2r_{o}}{c_{o}} \tag{7}$$

$$F_{\rm r} = \frac{\Delta F}{T} \tau \tag{8}$$

where  $c_o$  is the velocity of light. The resolution limit of the measured beat note frequency is determined by the time span T:

$$\Delta F_{\rm r} = \frac{1}{T} \tag{9}$$

If we have two targets with ranges  $r_1$  and  $r_2$ , their beat note frequencies can be estimated as, using equations (7) and (8):

$$F_{rl} = \frac{\Delta F}{T} \cdot \frac{2r_l}{c_o} \tag{10}$$

$$F_{r_2} = \frac{\Delta F}{T} \cdot \frac{2r_2}{c_o} \tag{11}$$

The resolution limit of the beat note frequency can be determined by:

$$\Delta F_{\rm r} = F_{\rm r2} - F_{\rm r1} = \frac{\Delta F}{T} \cdot \left(\frac{2r_2}{c_0} - \frac{2r_1}{c_0}\right) \tag{12}$$

Equating equations (9) and (12), the resolution limit of the measured range can be calculated as:

$$\Delta \mathbf{r} = \mathbf{r}_2 - \mathbf{r}_1 = \frac{\mathbf{c}_o}{2\Delta \mathbf{F}} \tag{13}$$

If a target is moving with a linear velocity  $v_r$  relative to the radar, a Doppler shift  $F_d$  is introduced which shifts the received signal in frequency. Figure 2 illustrates the received frequency compared to that of a stationary target.

Distance (range) r of an object moving with a relative velocity( $v_r$ ) at time t can be determined as:



Fig 2: Calculation of the velocity of a Moving target using the Doppler frequency shift

$$\mathbf{r} = \mathbf{r}_{\mathrm{o}} + \mathbf{v}_{\mathrm{r}} \cdot \mathbf{t} \tag{14}$$

Round-trip travel time is written as:

$$\tau(t) = \frac{2r_o}{c_o} + \frac{2v_r t}{c_o}$$
(15)

Using equation (6), phase of the beat note signal can be written as:

$$\varphi_{\rm D}(t) = 2\pi \left[ \frac{\Delta F}{T} \tau(t) \cdot t - \frac{1}{2} \frac{\Delta F}{T} \tau^2 + f_{\rm o} \tau \right]$$
(16)

Instantaneous frequency of the mixer signal:

$$f_{\rm D}(t) = \frac{1}{2\pi} \frac{\mathrm{d}\phi_{\rm D}}{\mathrm{d}t} \tag{17}$$

$$= \frac{\Delta F}{T} \left[ \tau + \frac{d\tau}{dt} \cdot t \right] - \frac{\Delta F}{T} \tau \cdot \frac{d\tau}{dt} + f_o \cdot \frac{d\tau}{dt} \quad (18)$$

Substituting for  $\frac{d\tau}{dt} = \frac{2v_r}{c_o}$ , the following is obtained:

$$f_{\rm D}(t) = \frac{\Delta F}{T} \cdot \tau + \frac{2v_{\rm r}}{c_{\rm o}} \cdot \left(f_{\rm o} + \frac{\Delta F}{T}(t-\tau)\right)$$
(19)

The first term is equal to  $F_r$  defined by equation (8) where  $\tau$  is approximately constant during measurement. Since the maximum time shift (t- $\tau$ ) is equal to T and  $\Delta F$  is typically much less than  $f_o$ , the second term is approximately equal to Doppler frequency  $F_d$ ; where:

$$F_{\rm d} = f_{\rm o} \frac{2v_{\rm r}}{c_{\rm o}}$$
(20)

So if there is a moving target, a Doppler frequency shift will be superimposed on the FM range beat note and an erroneous range measurement results. The Doppler frequency shift causes the frequency-time plot of the echo signal to be shifted up or down, as shown in Figure 2. On one portion of the frequency-modulation cycle, the beat frequency is increased by the Doppler shift, while on the other portion, it is decreased. If for example, the target is approaching the radar, the beat frequency produced during the frequency-increasing portion of the FM cycle will be the difference between the beat frequency due to the range  $F_r$  and the Doppler frequency shift  $F_d$ :

$$\mathbf{f}_{\mathrm{D1}} = \left| \mathbf{F}_{\mathrm{r}} - \mathbf{F}_{\mathrm{d}} \right| \tag{21}$$

Similarly, on the decreasing portion, the beat frequency is the sum of the beat frequency due to the range  $F_r$  and the Doppler frequency shift ( $F_d$ ):

$$\mathbf{f}_{\mathrm{D2}} = \left| \mathbf{F}_{\mathrm{r}} + \mathbf{F}_{\mathrm{d}} \right| \tag{22}$$

 $f_{\rm D1}$  and  $f_{\rm D2}$  are measured from which the range  $(r_{\rm o})$  and velocity  $(v_r)$  can be calculated simultaneously.

# 3. LINEAR FM CHIRP WAVEFORM GENERATION

The VCO is used in the system level simulations in order to produce the required transmitter signal. The VCO is modulated with a triangular source to produce a FMCW, as shown in figure 3. This chirp waveform, shown in figure 4, is used in high-resolution radars for small-target detection and target recognition. The radio frequency is increased at a constant rate from the carrier frequency  $f_o$  to  $f = f_o + \Delta f$  where  $\Delta f$  is the sweep bandwidth, as shown in figure 2, where T is the modulation period. This type of modulated waveform gets its bandwidth from the modulation and not the pulse width as for a waveform modulated with rectangular pulses whose spectrum has high frequencies. The chirp waveform bandwidth is given by the sweep bandwidth ( $\Delta f$ ). With this waveform it is possible to have wide bandwidths, giving good range resolution as indicated by equation (13), and wide pulse widths, giving a higher average radiated power and thus a higher probability of detection. For example, a sweep bandwidth  $\Delta f = 500$  MHz gives a range resolution of

$$\Delta \mathbf{R} = \frac{c_{o}}{2\Delta f} = \frac{3 \times 10^{8}}{2 \times 500 \times 10^{6}} = 0.6 \text{ m}$$
(23)

The linear modulation is continued for a period, (T), at least several times as long as the roundtrip transit time for the most distant target.



Fig 3: ADS schematics for linear modulations



Fig 4: VCO time domain signal



Fig 5: VCO Spectrum

In the VCO spectrum, shown in figure 5, the X-axis is not the absolute frequency, but the offset from the nominal analysis frequency, set at the carrier frequency. The actual spectrum of the VCO is shown in figure 6 which is used in the transient analysis. In both figures, it is obvious that the bandwidth is equal to the sweep bandwidth  $\Delta f = 100$  MHz.



Fig 6: VCO Spectrum using transient simulation

#### **4. DOPPLER TARGET MODEL**

The frequency of a signal reflected by a moving target is modeled. As explained previously, the moving target introduces Doppler frequency shift compared to a stationary target. If the transmitter's signal at an arbitrary frequency is given by:

$$\mathbf{E}_{t} = \mathbf{E}_{o} \cos(\omega t) \tag{24}$$

The echo signal from a moving target will be:

$$\mathbf{E}_{\mathrm{r}} = \mathbf{K}\mathbf{E}_{\mathrm{o}}\cos\left[\left(\boldsymbol{\omega} \pm \boldsymbol{\omega}_{\mathrm{d}}\right)\mathbf{t} + \boldsymbol{\phi}\right]$$
(25)

where

 $\mathbf{K}$ : Constant determined from the radar equation.,

 $E_{o}$ : Amplitude of the transmitter signal.

 $\omega$ : Angular frequency from transmitter, rad/s.

 $\omega_d$ : Doppler angular frequency shift.

#### $\phi$ : Constant phase shift.

The positive sign, in equation (25), indicates approaching target while the minus sign indicates diverging target. Equation (25) can be rewritten as:

$$\mathbf{E}_{\mathrm{r}} = \mathbf{K}\mathbf{E}_{\mathrm{o}}\cos\left[\left(\omega \mathbf{t} + \boldsymbol{\phi}\right) \pm \omega_{\mathrm{d}}\mathbf{t}\right]$$
(26)

Using the trigonometric identity

$$\cos(A \pm B) = \cos A \cos B \mp \sin A \sin B$$
,

equation (25) can be written as:

$$\mathbf{E}_{r} = \mathbf{K} \mathbf{E}_{o} \left[ \cos(\omega t + \phi) \cdot \cos \omega_{d} t \mp \sin(\omega t + \phi) \cdot \sin \omega_{d} t \right]$$
(27)

The phase shift  $\phi$  results from the transit time due to ranging.Equation (26) can be implemented in the target model, shown in figure 7, using HP-Advanced Design System (ADS). It can be also implemented in other SPICE-like circuit simulators. This model receives an incident signal and generates the corresponding reflected signal from a generally moving target. The incident is first delayed by the transit time calculated by equation 7. The signal is then divided equally by a power splitter. One of the outputs of the power divider passes through a phase-shifter with  $90^{\circ}$  phase shift while the other one goes unchanged. As a result, the outputs of the phase-shifters are quadratic to form the  $\cos(\omega t + \phi)$  and  $\sin(\omega t+\phi)$  terms. These signals are then multiplied by the corresponding quadratic signals  $\cos(\omega_d t)$  and  $\sin(\omega_d t)$ , respectively. The outputs of the multipliers are added up using a power combiner. The amplitude of the reflected signal is then controlled by an amplifier block to model the radar, environment, and target properties.

# 5. SYSTEM MODEL AND SIMULATION RESULTS

Figure 8 demonstrates the block diagram of a FMCW radar system model as implemented using ADS. It consists mainly of a transmitter and a receiver as well as a target model. The target model was described in the previous section. The transmitter path starts by a linear FM chirp waveform generator. The output of the generator goes into a power splitter in order to feed the mixer of the receiving path. The other output of the splitter is applied to a circulator from which it is directed to the antenna. The transmitted signal, by the antenna, hits the target, represented by an automobile symbol.

The reflected signal is modeled into the target, and then returned back to the antenna. The received signal, by the antenna, goes to the circulator from which it is directed to the LNA followed by a mixer. The LNA output signal is mixed with a portion of the transmitted signal to generate the beat note signal. The low frequency beat note signal is automatically filtered out in the mixer-IF block. The beat signal frequency is then analyzed to calculate the distance and velocity of the target.

Phase noise may be added to the VCO, via the phase noise modulator component, as shown in Figure 9. The phase noise modulator component adds phase noise in accordance to

to the transit time. The beat signal and its frequency are displayed in the third and second graphs from top, respectively. The maximum value of the beat frequency( $F_r$ ) can be estimated as, using equation (8):



Leeson's equation [15]:

$$L(f_{m})_{dB} = 10 \log_{10} \left[ FKT \left( 1 + \frac{f_{1}}{f_{m}} \right) \left( 1 + \left( \frac{f_{o}}{2Qf_{m}} \right)^{2} \right) \right]$$
(27)

where F is an empirical constant based on the oscillator,  $f_1$  is another constant, Q is the quality factor, K is Baltzmann's constant, T is the temperature in Kelvin, and  $f_m$  is the offset frequency from the carrier frequency  $f_o.$ 

#### 5.1.1 Stationary target

In the case of a stationary target, velocity is set at 0 Km/s while range is arbitrary chosen as 40 m. The sweep frequency  $\Delta f$  is equal to 100 MHz with a sweep period T = 0.5  $\mu$ s. It is worthy to mention that the sweep period is typically in the range of hundreds of microseconds. The value here is short, without loss of general, in order to increase the sweep frequency; and consequently reduce the simulation time as it is governed by the lowest frequency in the system. The transit time estimated in the target model is equal to:

$$\tau = 2 * 40 / 3 \times 10^8 = 267 \text{ns}$$

Figure 10 demonstrates the simulation results using our system model with the stationary target. On the top of the figure, baseband signals represent both the modulating signal of the transmitter and the demodulated signal of the receiver after the LNA. The two signals experience a time shift equal

#### Fig 7: Target model

which is exactly matching the simulated value, as shown in the graph; it is also confirmed by the spectrum of the beat signal shown in the fifth graph, from top. The VCO output signal and its spectrum are shown in the Fourth and sixth graphs. The bandwidth is almost equal to the sweep frequency 100 MHz.

The simulation results for the stationary target assure the sanity of the model. They also demonstrate how the received signal can be processed in order to ultimately measure the range of the target.

#### 5.1.2 Moving target

A moving target is considered with a velocity of 300 Km/s and a range of 37.5 m. The velocity is scaled up to a virtual velocity for any craft in order to be compatible with the short sweep period. Once again, the system is still general and these numbers are scaled in order to shorten out the simulation time without loss of generality. The same sweep frequency  $\Delta f$  of 100 MHz with a sweep period 0.5 µs is employed.



Fig 8: FMCW radar system model



# Fig 9: FMCW radar system model with phase noise added to VCO

The new transit time can be evaluated to 250 ns. The simulation results are shown in figure 11. The modulating and demodulated signals are shown on the top of the figure. The demodulated signal is shifted up indicating Doppler frequency shift. The Doppler frequency shift and range frequency can be calculated, using equations (20) and (8), respectively, as:

$$F_{d} = 2.4 \times 10^{9} \times \frac{2 \times 300 \times 10^{3}}{3 \times 10^{8}} = 4.8 \text{MHz}$$
$$F_{r} = \frac{100 \times 10^{6}}{0.5 \times 10^{-6}} \times 0.25 \times 10^{-6} = 50 \text{ MHz}$$



Fig 10: Stationary target with range 40 m

The second graph from top demonstrates the frequency of the beat signal with two distinct levels at the difference  $f_{D1} = 45.2$  MHz and sum  $f_{D2} = 54.8$  MHz. These results are in good conformity with the theoretical analysis. The spectrum of the beat signal, in the fifth graph, obviously has two peaks at almost same frequencies. In typical radars, these two frequencies can be used to estimate  $F_r$  and  $F_d$  from which velocity and range can be calculated. The VCO output signal and its spectrum are shown in the fourth and sixth graphs. The bandwidth is almost equal to the sweep frequency 100 MHz.



Fig 11: Moving target with a virtual velocity of 300 Km/s and a range of 37.5 m

# **5.2 Practical Considerations**

Some practical considerations can be inferred from simulations of our system model. These considerations impact the system level performance and should be taken into account during the distribution of the block level specifications. Some general guidelines for the system level designer can be used. Typical numbers of block level parameters differ from system to another. They depend on the whole system design, which is based on the technology and required specifications of the system.

### 5.2.1 Effect of rejection parameters of the mixer

System level simulations explain that rejection parameters of the mixer LO\_rej1 and LO\_rej2 should be large. LO\_rej1 and LO\_rej2 parameters are the local oscillator signal to input port rejection and output port rejection in dB, respectively. The reason is that low values of these rejection parameters result in higher values of DC component in the output spectrum. Such DC component can saturate subsequent block. On the other hand, the RF signal to output port rejection RF\_rej has no effect. This is can be justified by the low power of the reflected signal.

# 5.2.2 Effect of isolation parameters of the circulator

Circulator isolation from the transmitter port to the receiver port should be high. If the isolation is not large enough, some transmitted power can leak into the receiving path resulting in saturation of the LNA and subsequent blocks.

### 5.2.3 Effect of gain and noise-factor of LNA

The gain of the low-noise-amplifier LNA, indicated by  $S_{21}$ , should be large while its noise factor (NF) should be low. These are basic requirements for the LNA to get a low value for the overall NF of the receiving path. Specifically, a high gain for the LNA before the mixer is important to make it easy to resolve the weaker target echo.

The noise factor (NF) of the LNA has been swept from 3 dB to 21 with  $S_{21}$  fixed at 5 dB. Simulations demonstrated an evident change in the output for different values of NF.

# 6. CONCLUSIONS

A full RF front-end transceiver of a FMCW radar system model has been proposed and simulated. The simulation methodology has been implemented in ADS; however it is applicable to other SPICE-like circuit simulators. The model includes typical receiving and transmitting paths along with target model. The development of the target model has been provided. VCO phase noise has been also incorporated into the system model. The system model is flexible to adapt other realistic effects such as channel model. The channel model can be inserted between the antenna and the target model to accommodate for the attenuation and noise of the channel. The system has been simulated for stationary as well as moving targets. The simulation results showed good agreement with theoretical ones. Finally, some practical considerations were concluded for RF front-end system designers.

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