CFD and CAA Analysis of Centrifugal Fan for Noise Reduction

S. Ramakrishna
GVP College of Engineering (A)
Visakhapatnam

Vommi Krishna
ANITS
Visakhapatnam

A. Ramakrishna
A. U College of Engineering (A)
Visakhapatnam

K. Ramji
A. U College of Engineering (A)
Visakhapatnam

ABSTRACT
Centrifugal fans usually operate at high rotating speeds and hence generate high levels of noise whose reduction over a broad range of operating conditions is a key concern for the designer. An important first step in this direction is identification of noise sources and its quantification. For this purpose, Computational Fluid Dynamics (CFD) and computational aeroacoustic analyses is performed on centrifugal fans with backward impeller, radial impeller and forward impellers using CFD code FLUENT. To validate these results noise measurements are performed on centrifugal fans with backward impeller, radial impeller and forward impellers. The results so obtained from numerical analyses are compared with the measurement results. It is found that there is good agreement between numerical and experimental results. From these results it is observed that least noise of 92.6 dB(A) and high flow rate of 0.399 m³/sec occurs with backward impeller whereas large noise of 102.5 dB(A) and least discharge of 0.365 m³/sec is observed with forward impeller. The present study concludes that backward impeller radiates less noise and high discharge compared to radial and forward impellers.

Keywords
Centrifugal fan, backward impeller, radial impeller, forward impeller, CFD, aeroacoustics

1. INTRODUCTION
A Centrifugal fan is a roto-dynamic fan that uses a rotating impeller to deliver high flow rate with moderate pressure rise. Due to numerous applications in industries, researches on centrifugal fans and their noise related problems is increasing in recent times. Christopher [1] conducted experiments on three different types of fans to determine the major noise generation mechanisms. Wu [2] demonstrated that monopole exists for a fan running inside heating ventilation and air-conditioning unit of a vehicle. Sonetal [3] simulated the effects of bell mouth geometries on the flow rate of centrifugal fan numerically using a FLUENT. Chen-kang and Mu-En [4] used the CFD package FLUENT to simulate four backward impeller curved air foil centrifugal fans. Cao and Hu [5] proposed a cluster design approach to achieve a good aeroodynamic and acoustic performance of a ventilation system. Moreland [6] explained the housing effect on centrifugal fan for noise reduction. Lu et al. [7] presented a numerical optimization to reduce the vibration and noise of a centrifugal fan volute. The work of Scheit et al. [8] is concerned with analyzing the flow structures inside isolated radial impellers together with the far-field sound radiated from them in order to optimize the aerodynamic and acoustic performance. Prezelj and Cedina [9] concluded that the dominant noise source of a centrifugal fan can be attributed to the aerodynamically generated noise which exceeds the vibration-induced noise for more than 10 dB in a broad frequency range. Mao et al. [10] presented aeroacoustic tonal noise radiated from a centrifugal fan using a hybrid computational aeroacoustics approach that couples numerical flow with an implementation of the acoustic analogy. Mao et al. [11] calculated the noise radiating from rotating blades surrounded by a centrifugal volute by combining a hybrid computational aeroacoustic method with the thin-body boundary element. The present problem considers a single stage centrifugal fan consisting 12 blades driven by an electric motor. Experimental as well as numerical study on the centrifugal fan with three types of impellers has been carried out to evaluate their aeroacoustic and aerodynamic performance at optimum efficiency and corresponding operating speed of 2260 rpm.

2. EXPERIMENTAL STUDIES
2.1 Experimental setup
The Experimental setup consists of centrifugal fan driven by a/c motor through a step pulley arrangement to obtain different speeds through a step pulley arrangement to obtain three different speeds. The capacity of motor is 3.375 KW and all experiments are carried out at optimum speed 2260 rpm. The efficiency of motor is 80% and V-belt transmission efficiency 95% is consider for calculation of input power to a centrifugal fan. The motor is mounted on an adjustable bed and the fan output (discharge) is controlled by a sluice valve. All the measurements are carried out at optimum operating conditions to compare the noise and flow of centrifugal fan with backward, radial and forward impellers.

Figure 1 shows the experimental setup for aeroacoustic measurements. The noise of centrifugal fan was measured by Bluer & Kjaer sound level meter-2260, which is a precision hand-held sound level meter with real-time displays of 1/3-octave band in the range of 6.3 Hz - 20 kHz. Sound analysis software BZ7210 was used to analyze data obtained from sound level meter. The noise measured in room size of 5mx5mx3m. The background room noise is 47.2 dB(A) is considered for all noise measurements. The noise and flow rate of all backward, radial, forward impellers of centrifugal fan are compared at optimum rotational speed 2260 rpm. The sound pressure level difference between the centrifugal fan and background noise is beyond 10 dB, so the correction factor drops below 0.5 dB and therefore background noise becomes insignificant in present noise calculations. Figure 2 shows the experimental setup for aerodynamic measurements. An orifice meter is fixed in the outlet of pipe line to measure the actual discharge. A pitot tube and thermometer set is provided at the outlet to measure the velocity and temperature. U-tube manometers are provided to measure the pressure difference across the orifice meter and pitot tube.
2.2 Results and discussion
The noise levels of centrifugal fan with three different types of impellers namely backward impeller, radial impeller and forward impeller are measured and the results are compared at optimum operating conditions. Results are presented for the measurement point located at 1m upstream of the fan inlet, with the microphone aligned to the impeller shaft for all noise measurements. 1/3 octave band noise analyses are carried out for all impellers at optimum operating conditions. The sound pressure levels of all fans is measured at constant speed of 2260rpm.

The A-weighted sound pressure levels has been used since it reflects the noise perceived by the human ear more closely. Broadband analysis is used to obtain noise spectra in the range of 0-16.0 kHz. Figure 3 shows the 1/3 octave band sound pressure spectrum of backward impeller at inlet. These results show that tonal noise is predominant in low frequency range 0-1000 Hz, while broadband noise extends through the entire range of frequency 0-16.0 kHz. The overall sound pressure level for backward impeller is 92.6 dB(A). Figure 3 denotes peak value of 82.3 dB which is observed at fundamental blade passing frequency (452 Hz). Besides, another peak value of 78.8 dB is observed at first harmonic of blade passing frequency (904 Hz). Some of the noise peaks of blade passing frequency and their higher harmonics, could be due to resonant frequency of the noise generated by electric motor, vibration due to unbalanced rotating masses. Figure 4 shows the1/3 octave band sound pressure spectrum of radial impeller at inlet. The overall sound pressure level for radial impeller is 97.3 dB(A) which is 4.7 dB(A) more compared to backward impeller. Figure 5 shows the1/3 octave band sound pressure spectrum of forward impeller at inlet. It generates more noise when compared to backward impeller and radial impeller. The overall sound pressure level for forward impellers is 102.5 dB(A). It is 9.9 dB more than the noise generated by backward impeller and 5.2 dB more than that of radial impeller. Lesser noise generated by backward impeller when compared to radial impeller and forward impeller may be due to smooth flow at the inlet and outlet. Figure 6 shows the comparison of 1/3 octave band sound pressure spectrum of the three types of impellers. It shows that all fans generate peak noise between 450 Hz to 1000 Hz.

At the optimum operating condition the high discharge is observed for backward impeller is 0.399 m$^3$/sec and least discharge 0.365 m$^3$/sec is observed for forward impeller. Radial impeller discharge is 0.378 m$^3$/sec is in-between the discharge of backward impeller and radial impellers. Maximum pressure of 2079.5 N/m$^2$ and high velocity of 29.6 m/sec is observed for backward impeller. Low pressure of 1471.8 N/m$^2$ and least velocity of 25.66 m/sec is observed for forward impeller. From these results high discharge, velocity and pressure is observed with backward impeller and least velocity, discharge and pressure is observed with forward impeller whereas aerodynamic and aeroacoustic results of radial impeller is in between these impellers.
Figure 4. 1/3 octave band sound pressure spectrum of backward impeller at inlet

Figure 4. 1/3 octave band sound pressure spectrum of radial impeller at inlet

Figure 5. 1/3 octave band sound pressure spectrum of forward impeller at inlet
3. NUMERICAL STUDIES

3.1 Geometric Modelling

In the present problem reverse engineering methodology is used to obtain dimensions and three-dimensional model of centrifugal fan. Reverse engineering method is simply an effort to try and recreate the design of a product by examining the product itself. The impeller is major noise source compared to the casing. So in the present numerical simulations are carried out by taking the impeller as a noise source. Solid model of backward impeller, radial impeller and forward impellers is obtained using CATIA v5r10.

The outlet diameter of all impellers is 440 mm, inlet diameter is 200 mm and it contains 12 equally spaced blades. The minimum distance between the impeller and the casing is 55 mm. The widths of the impeller and casing are 100 mm and 210 mm respectively. Figure 7 show the solid model of backward impeller and Figure 8 show the solid model of computational domain used for present noise and flow simulations.

3.2 CFD and CAA Analyses

The objectives of the present Computational fluid dynamics and Computational aeroacoustics analyses are to find the noise sources and sound pressure level of centrifugal fan at optimum efficiency operating point. Commercially available grid generation code GAMBIT is used to mesh the domain of centrifugal fan. Three dimensional structural tetrahedral grids are generated to discretize the domain. In order to obtain better results tetrahedral element has been used by taking advantage of mesh consistency.

The domain is discretized by minimum 10 linear elements per source wavelength to increase the accuracy of the acoustic propagation. Different grids of element sizes 12, 10, 8 and 6 are generated for the entire computational domain of centrifugal fan with backward impeller. The simulations are carried out for all grids in Fluent for comparison of results. From the results it is observed that close results are observed with element size 8 and 6. So grid with element size 8 is used for noise and flow simulations of centrifugal fan with backward impeller, radial impeller and forward impellers. After mesh convergence with element size 8, there are 1.45 million tetrahedral elements are generated for the entire domain of centrifugal fan with backward impeller. Figure 9 show the mesh generated near the centrifugal fan with backward impeller and casing. It is clearly shows that denser mesh is near blade surface to capture the flow properties with significant quality. The similar meshes are also generated for centrifugal fan with forward and radial impellers.

The numerical simulations have been carried out with a finite volume code FLUENT. Three dimensional segregated implicit solver is used in the present analysis. Second-order implicit method is used for time discretization. The turbulent nature of the flow is incorporated through the Large Eddy simulation which is chosen as viscous model since it needs time dependent solution for aeroacoustic solution and is not highly dependent on geometrical conditions. Fluid zone in the volume is defined as “moving mesh” and rotate at speed of 2660 rpm in X-direction. Merged equations were solved iteratively at time increment, and set in such way that 10,000 increments correspond to one single revolution of the fan. The corresponding time step of $1.23148 \times 10^{-5}$ s for angular speed of 2660 rpm is used for the present simulations. The wall...
forming the centrifugal fan blade is assigned a relative rotational velocity of zero with respect to adjacent cell zone. A pressure inlet is prescribed at inlet and outlet, pressure condition is set at outlet of centrifugal fan. Computation is terminated when changes in solution variables from one iteration to the next is negligible. Fig. 10 shows the boundary conditions imposed on the computational domain of centrifugal fan.

3.3 Results and Discussion

Table 1 shows the comparison of simulation and measurement results of noise and flow rate of the three types of impellers. The noise source used for present simulations is only impeller and flow field in the impeller and volute is defined by the domain boundary conditions. For all impellers noise and flow parameters are simulated at same as experimental optimum operating conditions to compare the numerical and experimental results.

The receiver point is aligned to the impeller shaft and is located at 1 m upstream of the centrifugal fan inlet. This is a common location for microphones in the test setup. Figure 11 shows noise prediction graph of backward impeller at inlet. To observe broadband noise, frequencies from 0 Hz to 4,000 Hz are plotted. As shown in Figure 11 broadband noise extends over the entire range of frequency and the overall sound pressure level value obtained from numerical simulation is 85.8dB(A) at inlet for backward impeller, 88.2 dB(A) for radial impeller and 93.9 dB(A) for forward impeller. From these results it is observed that the blade of the fan is the main contributor to the aeroacoustic characteristics of the flow field. From the unsteady flow results it is found that the predicted flow rate for all fans is same as actual flow obtained from measurements. Non-uniform flow at inlet causes non-uniform aerodynamic forces on the blades as their angular positions change. This generates noise at blade passing frequency and its harmonics and is one of the major noise sources of centrifugal fan. The basic noise source in centrifugal fan is dipolar noise source as noise is generated due to aerodynamic pressure fluctuation. The sound pressure levels at inlet of the radial impeller and forward impellers centrifugal fan is shown in Figure 12 and Figure13 respectively. From these results it is observed that error between predicted and measured aeroacoustics and aerodynamic results of all centrifugal fans is within 15%.

Table 1 Aeroacoustics and Aerodynamic results comparison

<table>
<thead>
<tr>
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<th>Backward impeller</th>
<th>Radial impeller</th>
<th>Forward impeller</th>
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<tbody>
<tr>
<td>Overall SPL (dB(A))</td>
<td>Measure</td>
<td>Prediction</td>
<td>Error (%)</td>
</tr>
<tr>
<td></td>
<td>92.6</td>
<td>85.8</td>
<td>7.3</td>
</tr>
<tr>
<td>Pressure at outlet (N/m²)</td>
<td>2079.5</td>
<td>2320</td>
<td>11.56</td>
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<tr>
<td>Velocity (m/sec)</td>
<td>29.6</td>
<td>32.56</td>
<td>10.00</td>
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<tr>
<td>Discharge (m³/s)</td>
<td>0.399</td>
<td>0.453</td>
<td>13.53</td>
</tr>
<tr>
<td>Total Efficiency</td>
<td>60.35</td>
<td>67.54</td>
<td>11.9</td>
</tr>
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</table>
Figure 11 Noise prediction graph of backward impeller at inlet

Figure 12. Noise prediction graph of radial impeller at inlet

Figure 13. Noise prediction graph of forward impeller at inlet
4. CONCLUSIONS
The aerodynamic and aeroacoustic parameters of centrifugal fan with backward impeller, radial impeller and forward impellers are predicted and plotted and results obtained from numerical analyses are compared with the measurement results. Close correlation is observed between experimental and numerical results. From these results it is concluded that centrifugal fan with backward impeller generates less noise and high discharge compared to radial and forward impellers. Least discharge and noise is observed with forward impeller whereas aerodynamic and aeroacoustic results of radial impeller is in between backward and radial impellers.

5. REFERENCES